

ABSTRACT:

The invention relates to a rotor for a gas turbine.

The rotor comprises several blades (12a, 13a, ...29b, 30b) rotating together with the rotor (10), particularly running blades, the blades (12a, 13a, ...29b, 30b) forming at least one blade ring.

According to the invention, the blades (12a, 13a, ...29b, 30b) within the or each blade ring are arranged at a different distance from one another.

(Figure 1)